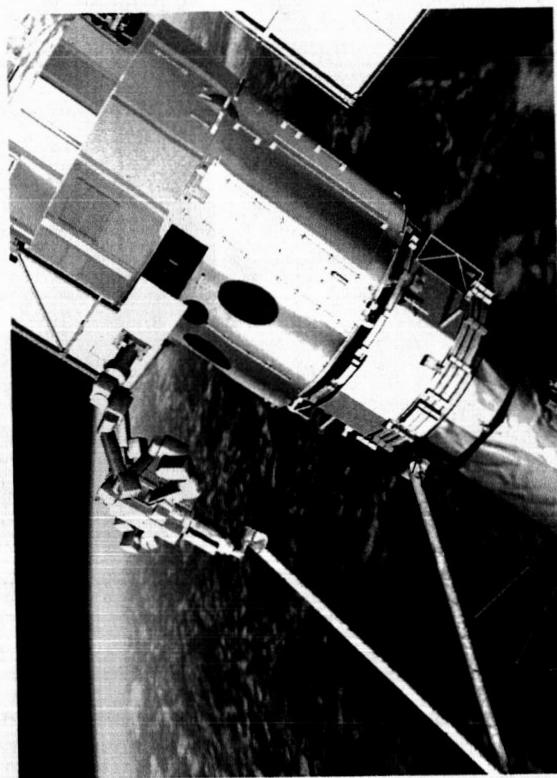
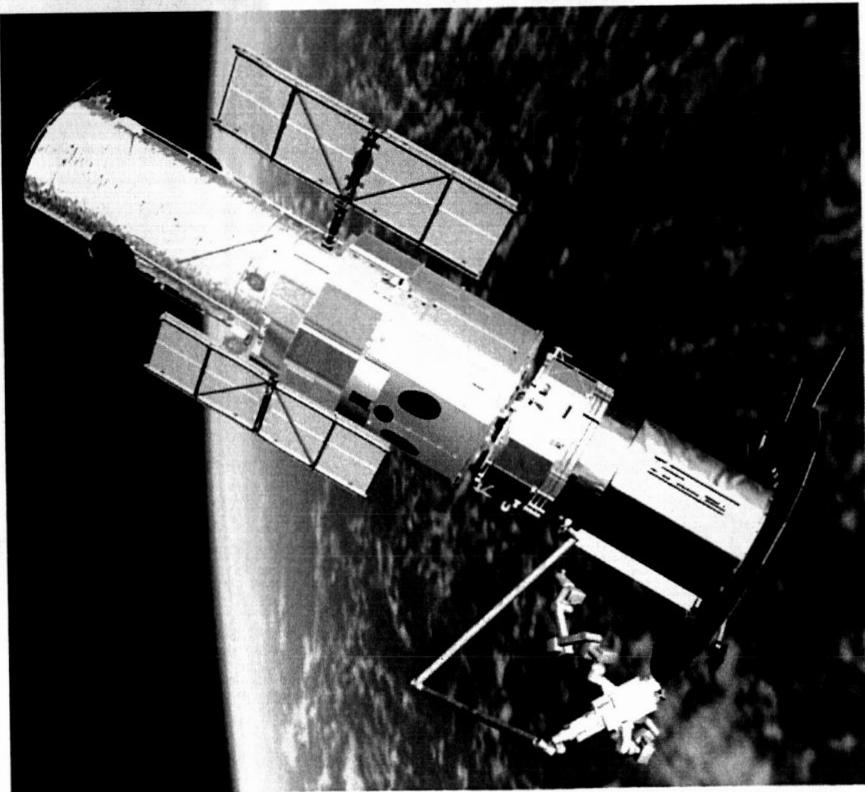




# Wide Field Camera 3 Accommodations for HST Robotics Servicing Mission



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# Acknowledgements

David Hughes

Wanda Peters

Jackie Townsend

Jack Triolo

# Overview

- HRSDM
  - Mission Objective
  - Hubble Robotic Vehicle
- What is Wide Field Camera 3 (WFC3)?
- Contamination Accommodations for WFC3
  - RSU analysis
  - IR Vent Tube



# HST Robotics Servicing and De-orbit Mission (HRSSDM) Objectives

- Provide a disposal capability at the end of HST's useful scientific life
  - The De-orbit Module (DM) will remain attached to the aft bulkhead on HST after servicing is complete
- Upgrade hardware by installing two new scientific instruments
  - Replace the Corrective Optics Space Telescope Axial Replacement (COSTAR) with the Cosmic Origins Spectrograph (COS)
  - Replace Wide Field / Planetary Camera-2 (WFPC2) with Wide Field Camera-3 (WFC3)
- Extend scientific life of HST for a minimum of 5 years after servicing



# HRV Element Functionality (1 of 2)

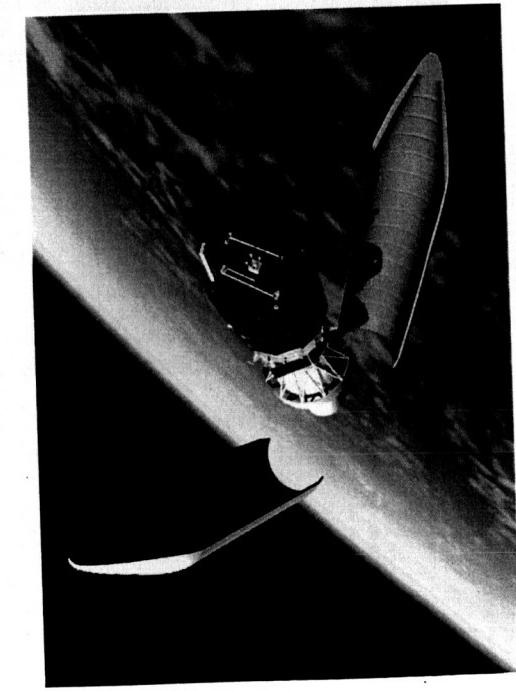
- Hubble Robotic Vehicle (HRV) is made up two spacecrafts and two robotic elements
  - The two spacecrafts are the De-orbit module (DM) and Ejection module (EM)
    - De-orbit module (DM)
      - Provide sensors and intelligence to rendezvous with HST
      - Support life extension activities
      - Provide disposal capabilities of HST at the end of its useful life
    - Ejection module (EM)
      - Houses all elements which are not required after robotic servicing
      - New science instruments are stored inside EM
      - Robotic elements are housed in or mounted on EM



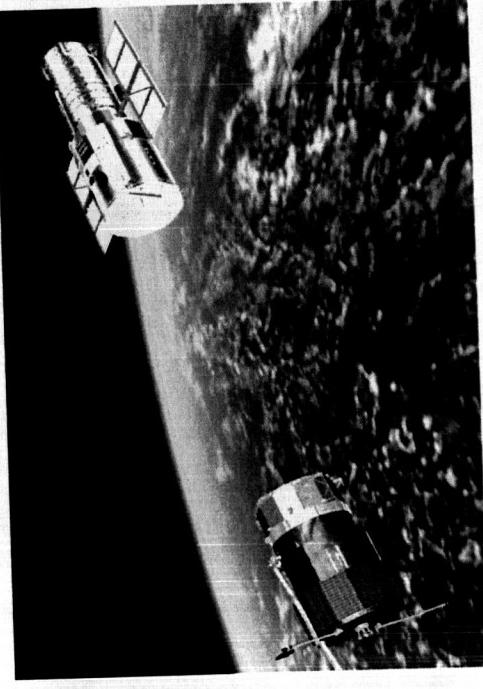
## HRV Element Functionality (2 of 2)

- The Robotic System is made up of two elements: Grapple Arm (GA) and Dexterous Robot (DR)
  - Grapple arm is used to capture HST
  - Dexterous robot is used to handle payload elements and perform servicing tasks

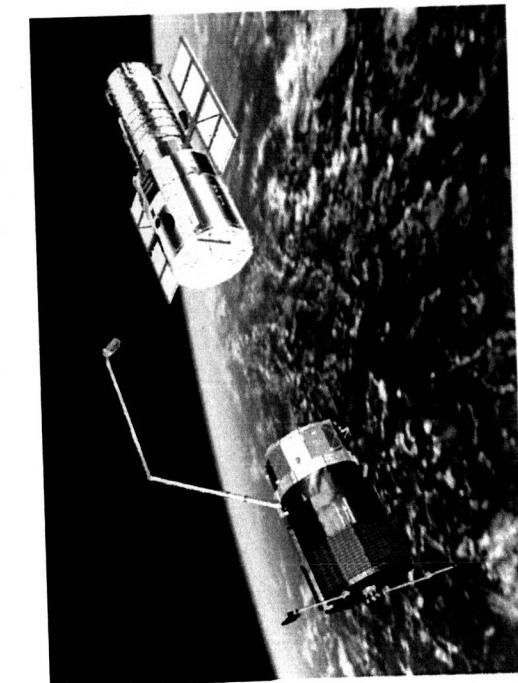
# HRV Mission Overview (1 of 2)



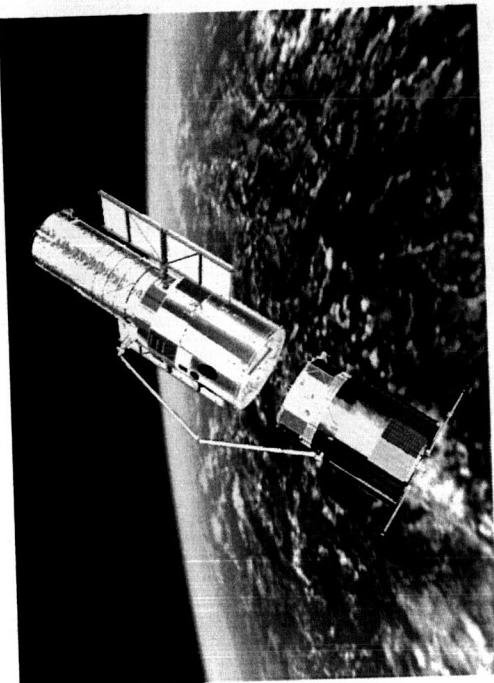
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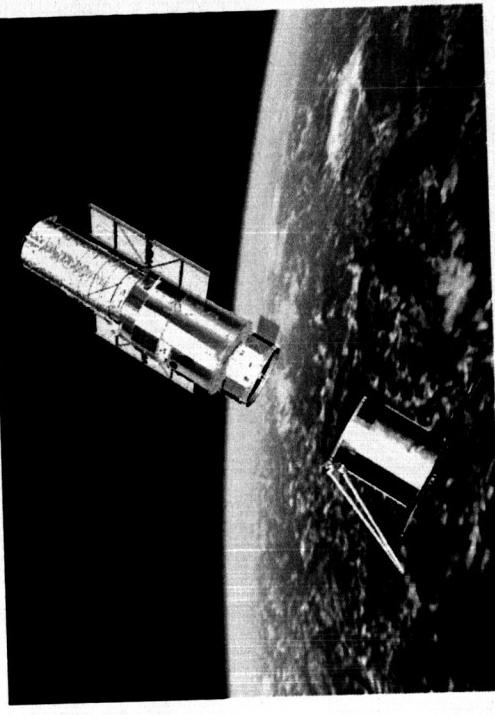
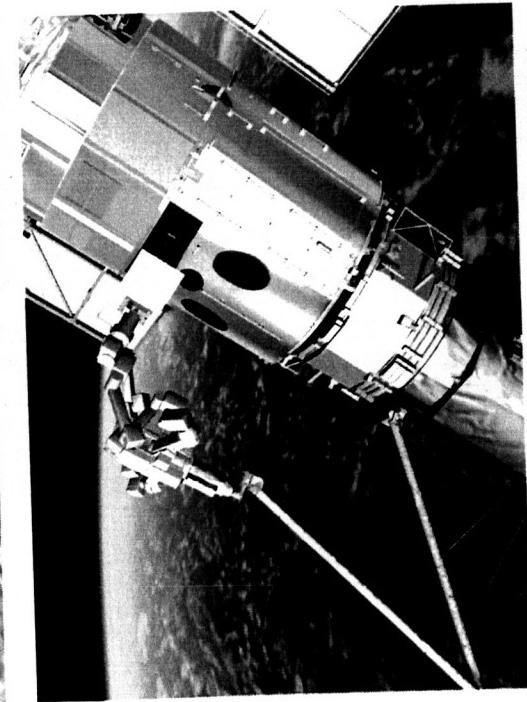
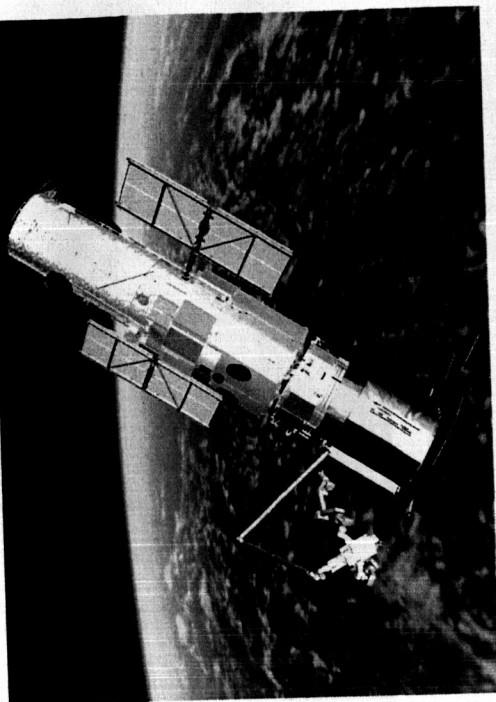
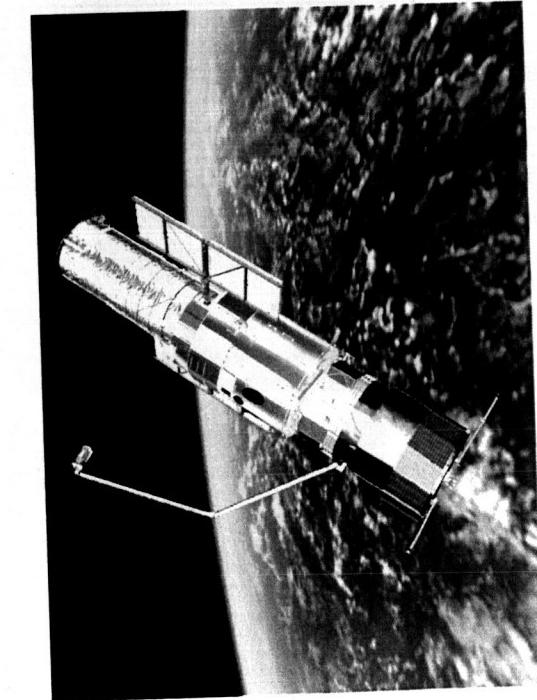
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4

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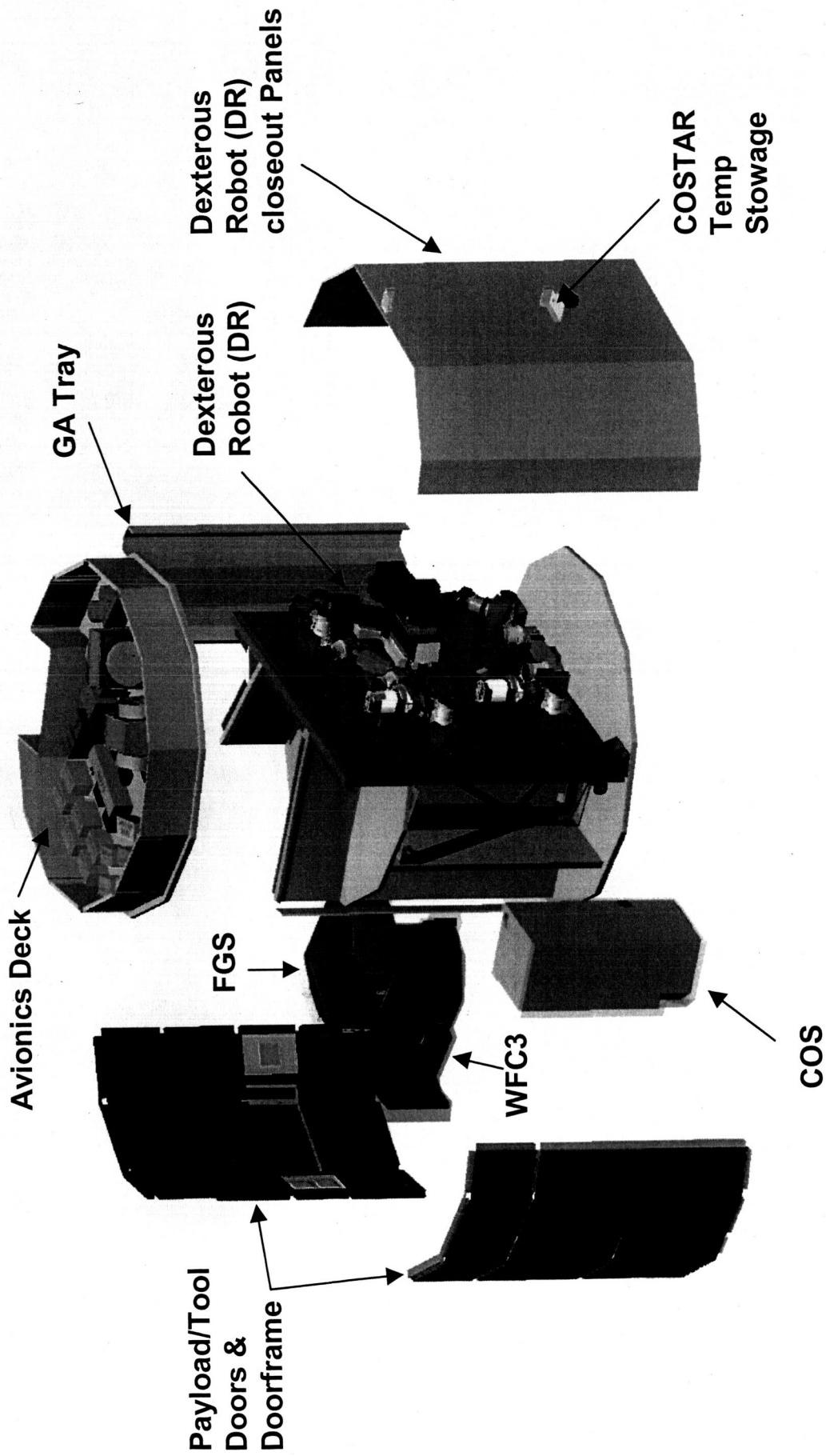
# HRV Mission Overview (2 of 2)



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# Hubble Robotic Vehicle

## EM Robot and Instrument Module



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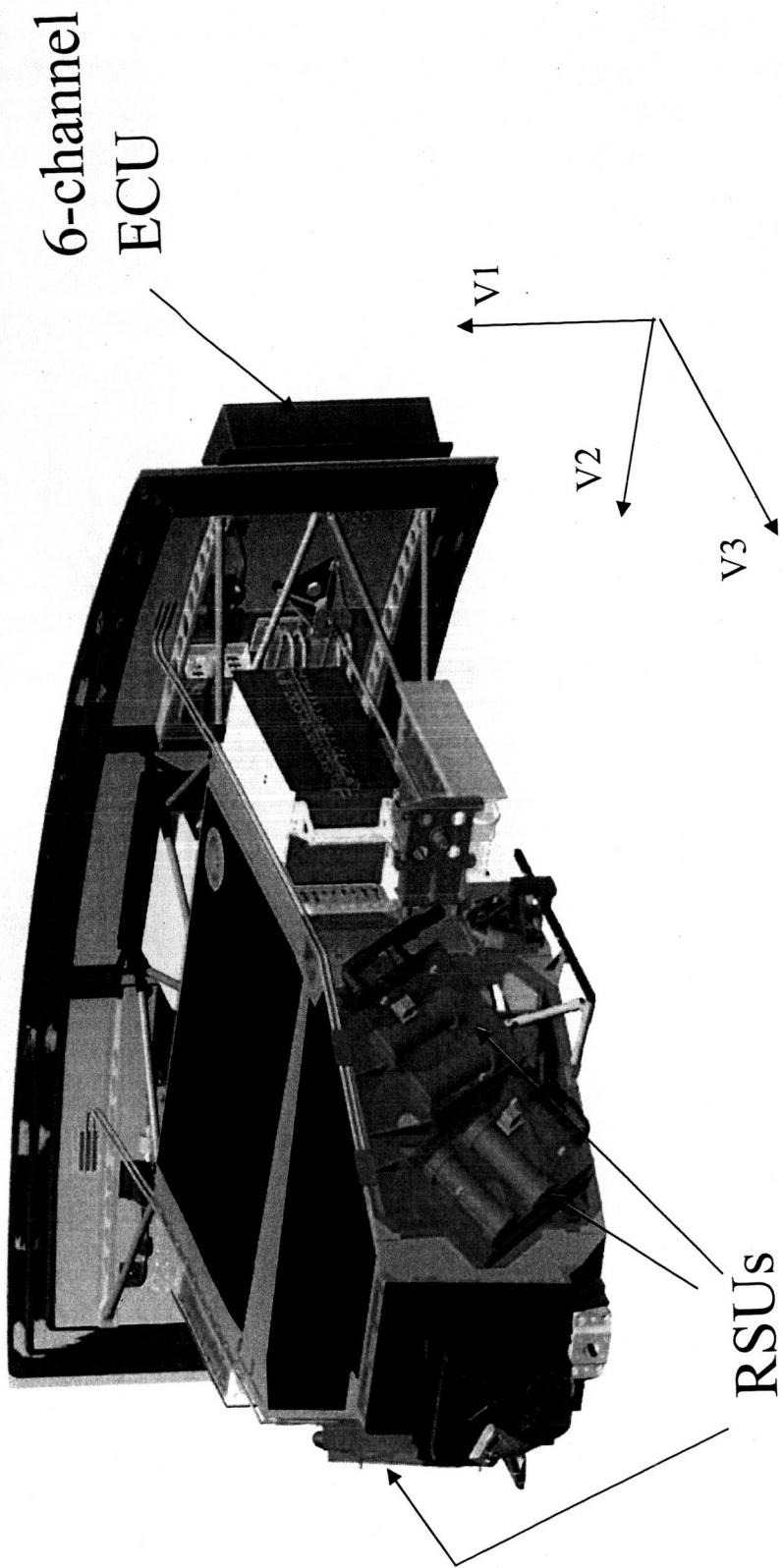
# What is WFC3?

- Provide HST with high quality imaging capability until the end of the HST mission
- WFC3 will be the first “panchromatic” camera on HST
  - Two channels cover Near-UV to Near-IR (200-1700 nm) without cryogen
- Facility-class scientific instrument
  - Built for the benefit of the HST user community, with no guaranteed observing time for the developers.

# Wide Field Camera 3

- WFC3 will provide the HST replacement gyros
  - The Rate Gyroscope Assembly (RGA) II is made up of one Electronic Control Unit (ECU) to support (3) Rate Sensor Units (RSUs)
    - Total of 6 Gyros housed in 3 RSU “boxes”
    - The (3) RSU boxes will be attached to the front diagonal panels of the WFC3 Enclosure and the ECU will be on the external Radiator
  - RSU mounting platforms will meet the Pointing Control requirements
    - 3.6 arc-sec motion
    - 2.4 arc-sec/hour rate

# Wide Field Camera 3

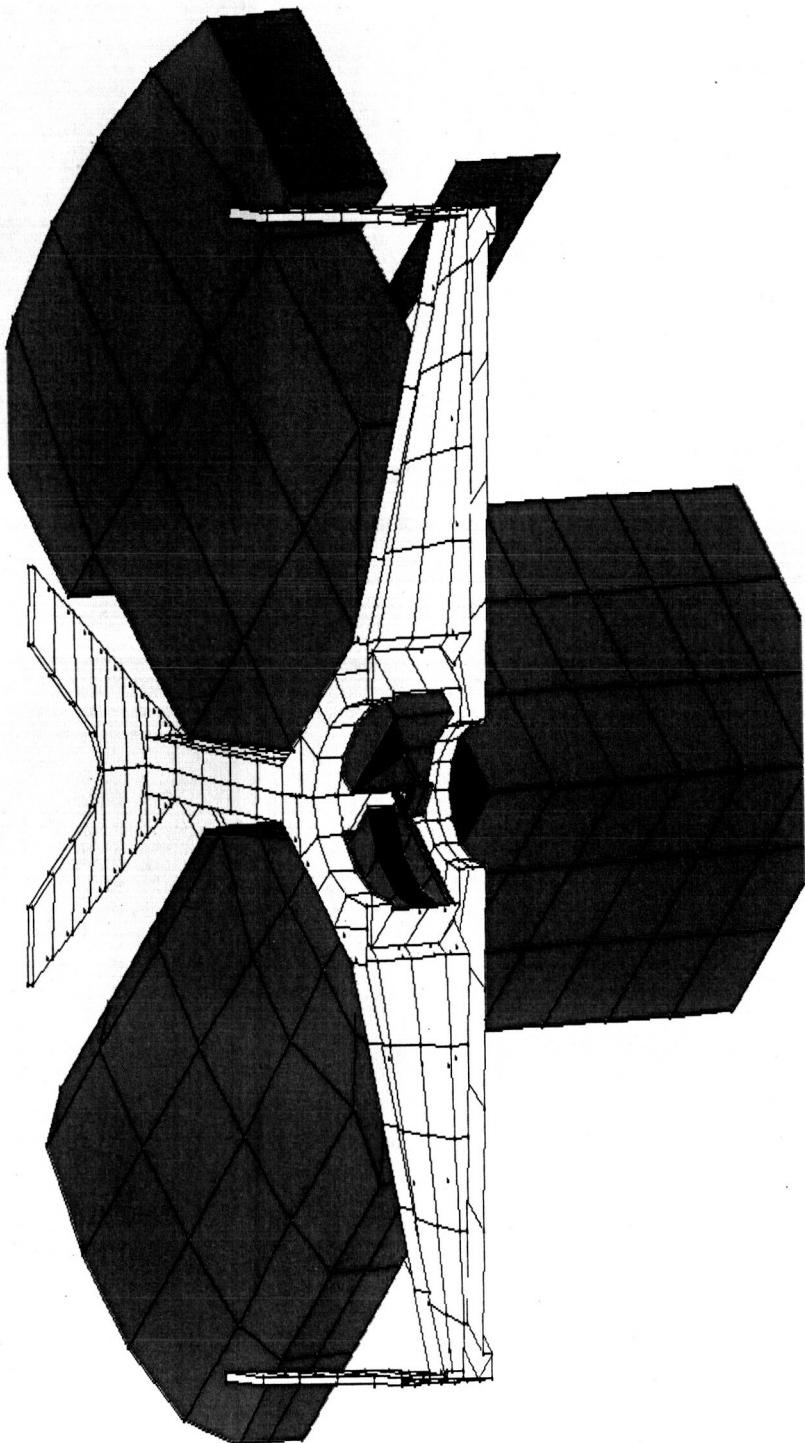


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# Contamination Accommodations for RSUs (1 of 4)

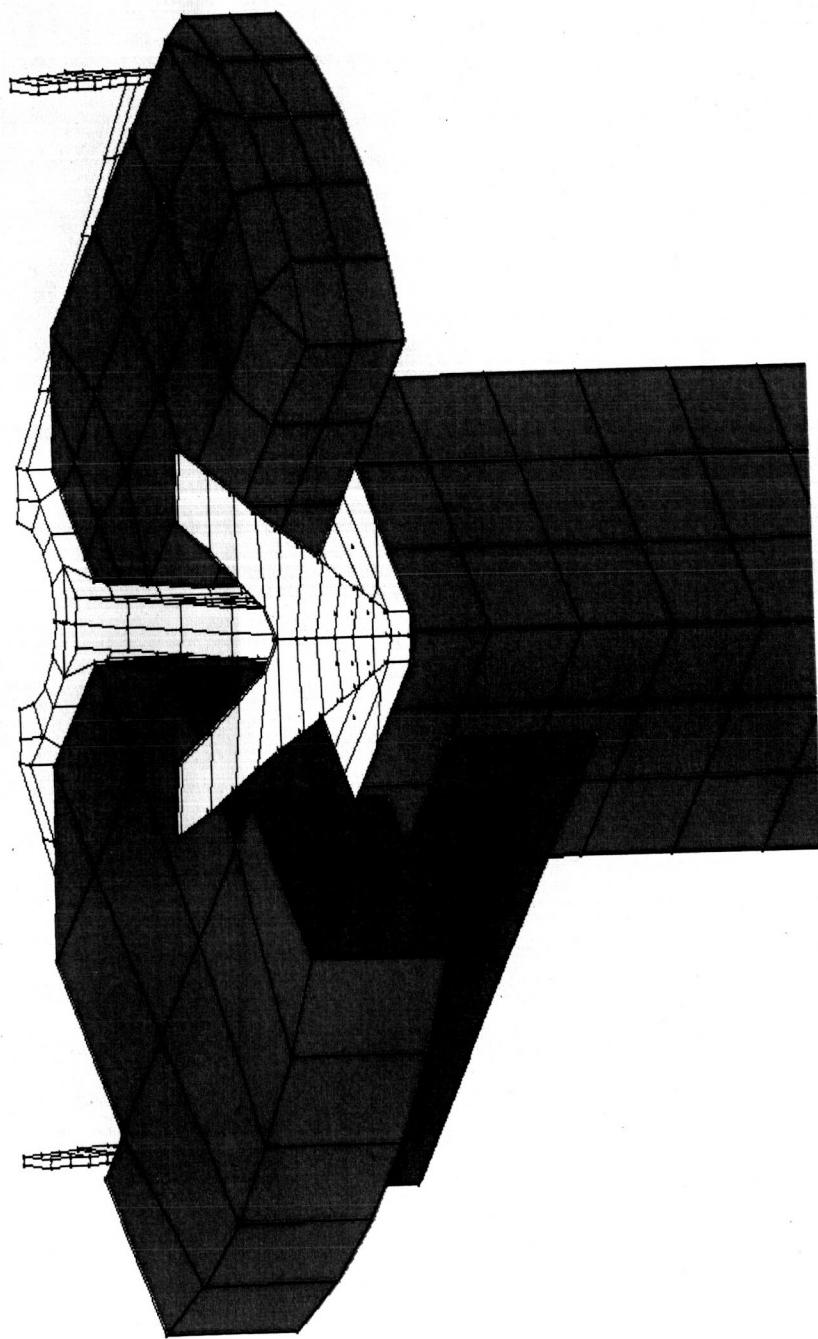
- Previous location of RSU was on the shelf below WFC3
  - Outgassing requirement of Aft Shroud:  $4.33 \times 10^{-13}$  g/cm<sup>2</sup>/sec
  - Actual outgassing rate of  $5.4 \times 10^{-13}$  g/cm<sup>2</sup>/sec achieved
  - Aft Shroud requirement waived by analysis
- New RSU box locations are on the sides of WFC3
  - Closer to optics
  - Analysis required to determine if actual outgassing rate is still acceptable

# Contamination Accommodations for RSUs (2 of 4)



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Contamination Accommodations for  
RSUS (3 of 4)



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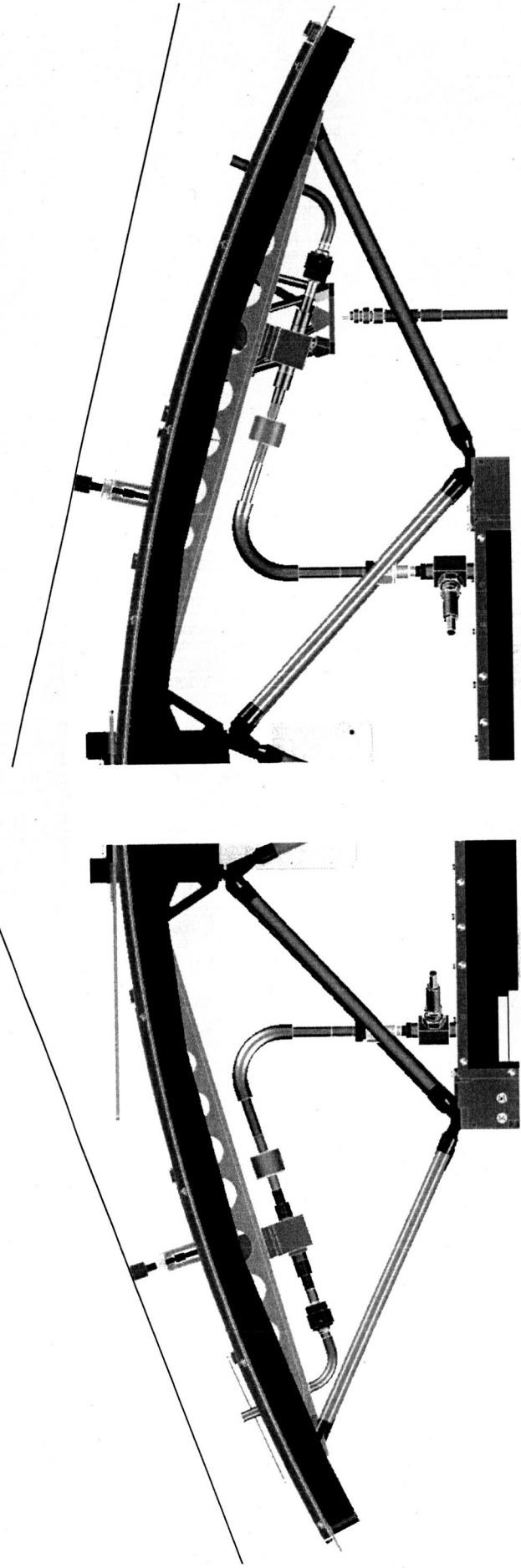
# Contamination Accommodations for RSUs (4 of 4)

- Analysis of previous location of RSU
  - On the shelf below WFC3
  - RSU measured outgassing rate on previous missions:  $5.4 \times 10^{-13}$  g/cm<sup>2</sup>/sec
  - Analysis predicted 0.01 Å accretion on WFC3 Pick-off Mirror per year
  - Results were acceptable
- Analysis of new RSU box locations
  - Attached to the front diagonal panels of the WFC3 Enclosure
  - Mr. Tony Dazzo and Mr. Dave Hughes performed analysis and modeled HST using IDEAS/TMG
  - Assume same outgassing rate achieved
  - Worst case deposition was 0.015 Å accretion
  - Results were acceptable

# Contamination Accommodations for IR Vent Tube (1 of 2)

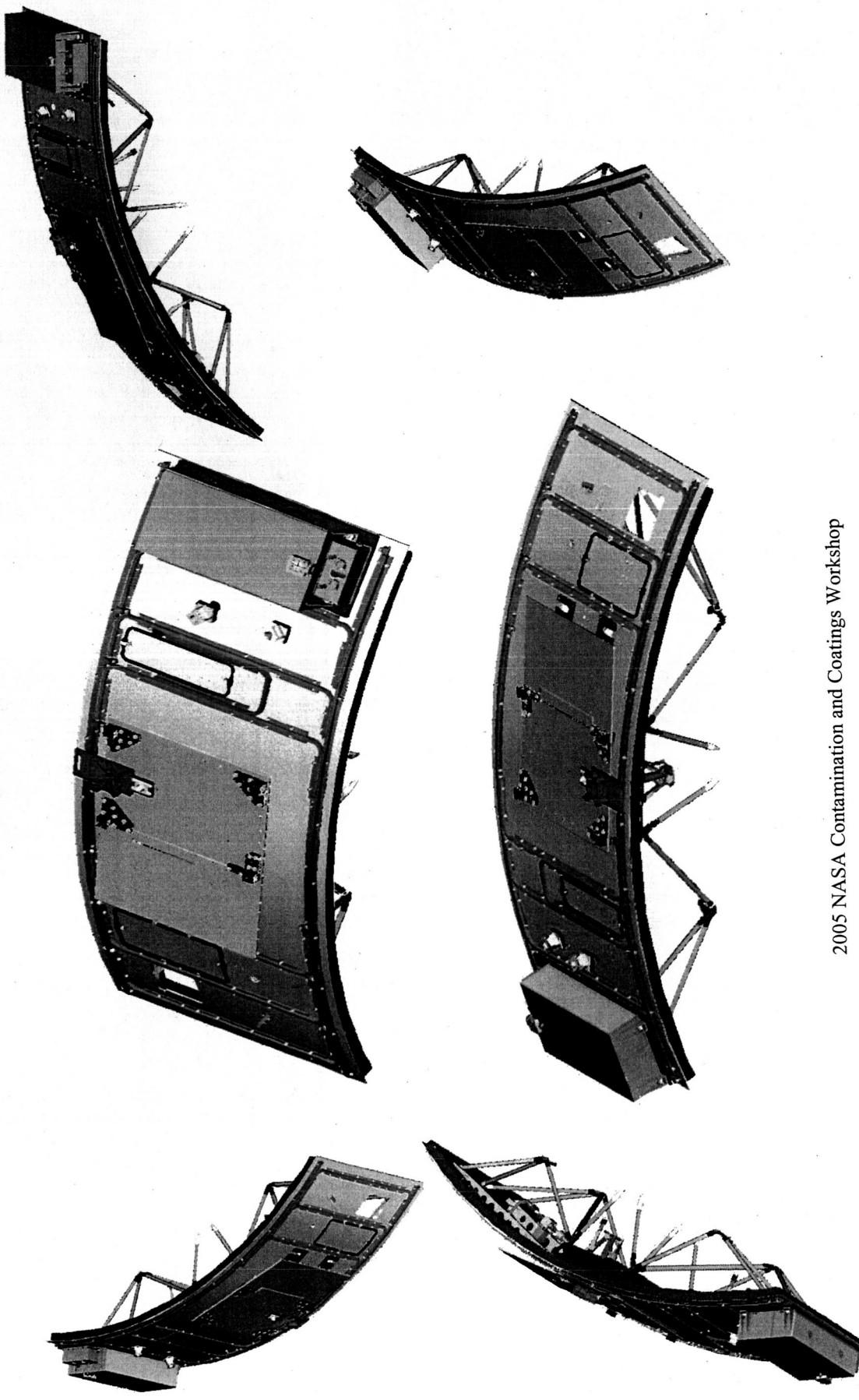
- IR Detector Vent Port Location
  - Pre-RGA II requirement: No object in line of sight of vent tube opening
  - New ECU box became the highest point on the -V2 side of radiator which is in line of sight
    - Vent tube opening in line of sight allows:
      - Exposure to contaminants and reflected outgassing to enter

No line of sight to vent tubes



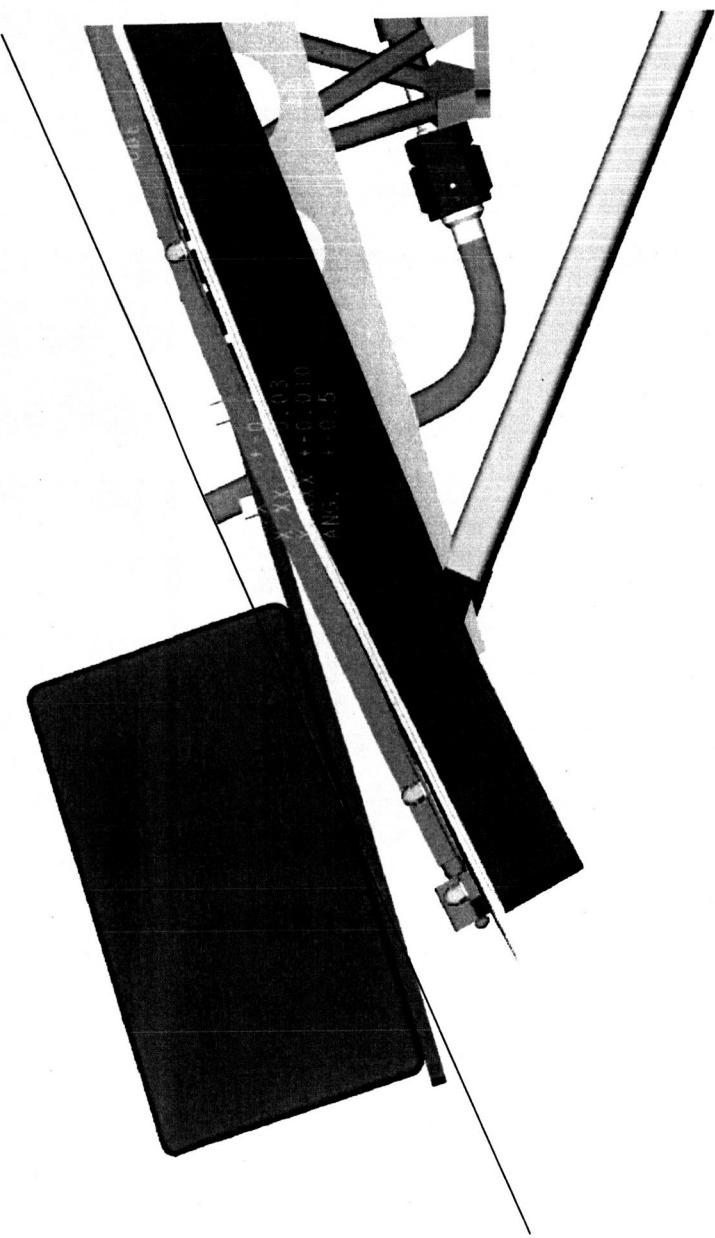
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# 6-Channel ECU Box Mounted on WFC3



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# ECU in line of sight



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# Contamination Accommodations for IR Vent Tube (2 of 2)

- Possible solutions to avoid line of sight to vent port
  - Place molecular adsorbers near port opening
    - Functions properly until maximum capacity has been reached
  - Extend the length of vent tube beyond height of ECU box
    - Not suitable due to constraint of robot arm clearance
  - Move detector port location to other side of radiator (UVIS side)
    - Needed to re-analyze the time for venting of tube
      - Analysis was performed by Dr. Michael Woronowicz
        - Time for IR detector to reach required pressure: 157 hrs. (old), 338 hrs. (new)
    - Both time results are within 3-week design goal and acceptable



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BALL

MEGA

RITSS